Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved	Electronic Tracking Number
OMB No. 2120-0020	
11/30/2007	
	or FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a)) Serial No. Nationality and Registration Mark N547SH 0752 USA 1. Aircraft Model Series Make 22 SR-22 Cirrus Address (As shown on registration certificate)

Address C/O NRAI 160 Greentree Dr. Ste 101 Name (As shown on registration certificate) Lago Sunrise LLC State DE 2. Owner Dover City Country USA 19904 3. For FAA Use Only The alteration or repair identified herein complies with the applicable airworthiness requirements and is soroved for use only on the above described aircraft, subject to conformity "Alusho inspection by a person authorized in § 43.7. APR 2 5 2008 District Office FAA Inspector 5. Unit Identification 4. Type Model Serial No. Make Unit Alteration Repair (As described in Item 1 above) X AIRFRAME **POWERPLANT PROPELLER** Type APPLIANCE Manufacturer 6. Conformity Statement B. Kind of Agency A. Agency's Name and Address U. S. Certificated Mechanic Manufacturer Aurora Aviation Address 355 McGregor Airport Road C. Certificate No. Foreign Certificated Mechanic TX Certificated Repair Station Mc Gregor City Country USA Certificated Maintenance Organization 76657 Zio D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Signature/Date of Authorized Individual Extended range fuel per 14 CFR Part 43 App. B April 25, 2008 7. Approval for Return to Service was inspected in the manner prescribed by the Pursuant to the authority given persons specified below, the unit identified in item 5 Rejected Administrator of the Federal Aviation Administration and is | X | Approved Persons Approved by Canadian FAA Flt. Standards Maintenance Organization Manufacturer Department of Transport Inspector ΒY Other (Specify) Х Inspection Authorization FAA Designee Repair Station Signature/Date of Auth Certificate or Desig April 25, 2008

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

USA N547SH

04/25/2008

Nationality and Registration Mark

Date

Aircraft S/n: 0752

Removed existing H.I.D. Ballast Assembly (Cirrus P/n: 12580-104) and harness.

Fabricated bracket, 4 inch by 6 inch, using .063" 2024-T3 aluminum sheet. Installed Aero Visions International Inc. XeVision HID Landing Light Ballast to the bracket using three MS24694S51 screws, NAS 1149-363P washers, and three MS21045-3 nuts. Installed assembly to the firewall using pre-existing holes and three MS27039-1-10 screws, and three NAS1149-363P washers. Fay sealed bracket using Dow Corning Red RTV #736.

The original Cirrus bulb assembly was used, P/n: 12580-123.

Installation performed in accordance with the XeVision Installation and Operation Manual and AC 43.13-1B.

Equipment list updated. Details on file under Work Order #: AA08-0091.

Instructions for continued airworthiness:

- 1) Introduction: Aero Visions International Inc. XeVision HID Landing Light System
- 2) Description: This assembly replaces the original landing light ballast and wire harness. The pre-existing lamp and mounting locations for the OEM system were used.
- 3) Control Operation Information: When bench testing the lamp, do not start, or shut down the lamp with the light beam oriented at an angle of more than 45 degrees above or below horizontal during startup (1st minute of operation) or cool down (1st minute after turn off). If the lamp is turned off and left in the vertical or near vertical position during the cooling cycle, METAL Halide salts (Yellow looking material in the center of the arc tube) will form, rendering the lamp inoperable and will ruin the bulb.
- 4) Servicing Information: N/a
- 5) Maintenance: For repairs or replacement contact Aero Visions International Inc. 4245 Airport Road, Ogden UT 84405. Website: www.xevision.com. Phone: (801) 622-7000.
- 6) Troubles hooting information: See attached wiring diagram and contact Aero Visions International Inc. 801-622-7000.
- 7) Removal and replacement information: Refer to XeVision Installation and Operational Instructions.
- 8) Diagrams: See Attached wiring diagram.
- 9) Perform Inspection annually in accordance with FAR 43, Appendix D.
- 10) Application of protective treatments: None.
- 11) Data: N/a
- 12) List of Special Tools: None.
- 13) N/A

X Additional Sheets Are Attached

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	USA N547SH	04/25/2008		
	Nationality and Registration Mark	Date		
14) Recommended overhaul periods: No Additional overhaul time limitations.				
15) Airworthiness limitation section: No Additional.				
16) Revision: A letter will be submitted to the local FSDO with a copy of the revised FAA Form 337 and revised ICA.				
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Additional Sheets Are Attached				