

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	GRUMMAN AMERICAN			Model AA-5A					
	Serial No. AA5A0487			Nationality and Registration Mark N26256					
2. Owner	Name (As shown on registration certificate) HARRIS NORMAN C			Address (As shown on registration of 702 MERRICK DR SUGAR LAND TEXAS 77478-3705			rtificate)		
		ein complied with ap			nly			***************************************	
above by a po APPRO	described aircraft erson authorized VING /INSPEGTO	ts and is approved t subject to conformi in FAR - 98.7.	only ty ins	pection	on			5. Type	
	T		-						
Unit	Make			Model Serial No.				Repair	Alteration
AIRFRAME	(As described in Item 1 above)								×
POWERPLANT									
PROPELLER									
	Туре								
APPLIANCE	Manufacturer								
			6.	Conformity State	ment				
A. Agency's N	ame and Address		B. Kind of Agency			C. Certifi	cate No.		
JERRY B. PARKER				X U.S. Certificated Mechanic					
919 MCKEEVER ROAD				Foreign Certificated Mechanic			A0D 40	ECE 40000	
ARCOLA, TEXAS 77583				Certificated Repair Station					
				Manufacturer					
		Iteration made to the unit(s Part 43 of the U.S. Federal							School State of the State of th
MARCH 2, 2007				Signature of Authorized Individual					
THE	17 x, 200		7 000	roval for Return T	o Service				
Pursuant to t	he authority given	persons specified bele iation Administration a	ow, the			ected in the man	ner prescr	ribed by the	
FAA	Fit. Standards ector			Inspection Authorization Other (Specify)		)			
FAA	Designee	Repair Station		Person Approved Canada Airworthir	oy Transport ess Group				
Date of Approval or Rejection Certificate or Designation No.				Signature of Authorized Individual					
MARCH 2	The state of the s	//	Ele -	tranic Version (Adaba)					

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the appicable airworthiness requirements.

8. Description of Work Accomplished

N26256 SN AA50487

REMOVED EXISTING PAR 36 LANDING LIGHT FROM THE COWL. INSTALLED XEVISION HIGH INTENSITY DISCHARGE (HID) LIGHT ASSEMBLY (KIT # XV-36-SL) INTO THE VACATED POSITION. THE BALLAST WAS MOUNTED ON THE FIREWALL. THE EXISTING SWITCHES, WIRING AND CIRCUIT PROTECTION WERE REUSED. THE PROVIDED HID WIRE HARNESS FROM THE BALLAST TO THE LIGHT WAS ROUTED DIRECTLY AND SECURED FOR CHAFE PROTECTION. THE AVERAGE AMPREE DRAW IS 4.2 AMPS..ALL WORK WAS PERFORMED IN ACCORDANCE WITH AC43.13-1B CH. 11 (SEC 3) PAR 11-31, 32, 37 (SEC 4) PAR 11-48 AND XEVISION INSTALLATION INSTRUCTUONS. AMENDED AIRCRAFT WEIGHT & BALANCE AND EQUIPMENT LIST.

----- END -----

## INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

THIS DOCUMENT IS TO BE ATTACHED TO FAA FORM 337, DATED 03-02-07 FOR THE AIRCRAFT LISTED BELOW.

MAKE

GRUMMAN

MODEL

AA-5A

SERIAL NUMBER REGISTRATION AA5A0487 N26256

THE INFORMATION CONTAINED HEREIN PERTAINS TO THE CONTINUED MAINTENANCE FOR THE MAJOR ALTERATION DESCRIBED IN ITEM 1 BELOW.

1. DESCRIPTION:

INSTALLED XEVISION LANDING LIGHT FOR INCREASED LIGHT ILLUMINATION AND FOR INCREASED SERVICE LIFE.

2. CONTROL OPERATION:

THE LIGHT IS CONTROLLED WITH THE EXISTING AIRCRAFT LANDING LIGHT SWITCH, THERE ARE NO SPECIAL PROCEDURES.

3.

3. SERVICING INFORMATION:

THE COMPONENTS ARE NOT FIELD REPAIRABLE AND MUST BE REPLACED WITH APPROVED COMPONENTS.

4. MAINTENANCE INSTRUCTIONS:

THIS LIGHTING SYSTEM IS TO BE MAINTAINED IN ACCORDANCE WITH FAR PART 43.13. INSPECTIONS ARE TO BE PERFORMED IN ACCORDANCE WITH FAR PART 43.15.

5. TROUBLESHOOTING INFORMATION:

IF THE CIRCUIT BREAKER POPS, REPLACE THE BALLAST UNIT. IF THE LIGHT DOES NOT ILLUMINATE, REMOVE THE LAMP OR BALLAST, VERIFY FUNCTION AND REPLACE AS REQUIRED. BENCH TESTING MUST BE DONE IN ACCORDANCE WITH THE XeVISION INSTALLATION AND OPERATION INSTRUCTIONS WHICH CONTAIN WARNINGS FOR BENCH TESTING.

6. REMOVAL AND REPLACEMENT INFORMATION:

THE HID LAMP IS REMOVED AND REPLACED IN THE SAME MANNER AS THE ORIGINAL. THE BALLAST IS ATTACHED TO THE FIREWALL USING STANDARD PROCEDURES AND HARDWARE. THE WIRE HARNESS BETWEEN THE BALLAST AND THE LAMP SHALL BE REMOVED IN ACCORDANCE WITH THE XeVISION INSTALLATION AND OPERATION INSTUCTIONS.

DIAGRAMS:

ACCESS IS THOUGH THE REMOVAL OF THE COWL, NO DIAGRAMS

8. SPECIAL INSPECTION REQUIREMENTS:

N/A

9. APPLICATION OF PR

N/A

10. SPECIAL HARDWARE:

11. SPECIAL TOOLS:

N/A

12. COMMUTER CATEGORY AIRCRAFT:

N/A

13. RECOMMENDED OVERHAUL PERIODS:

N/A

14. AIRWORTHINESS LIMITATIONS:

THERE ARE NO ADDITIONAL AIRWORTHINESS LIMITATIONS.

15. REVISIONS:

TO REVISE THESE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, A LETTER WILL BE SUBMITTED TO THE LOCAL FLIGHT STANDARDS DISTRICT OFFICE WITH A COPY OF THE REVISED FORM 337 AND REVISED ICA.

16. RECORD KEEPING:

THESE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS ARE TO BE PLACED IN THE AIRCRAFT PERMANENT RECORDS AND REFERRED TO DURING AIRCRAFT SYSTEMS INSPECTIONS AND MAINTENANCE.