MAJOR REPAIR AND ALTERATION-

Form Approved OMB No. 2120-0020

of Transportation Federal Aviation Administration		frame, Powerpl					Office Ident		
and dispositi	on of this form.	ype all entries. See FAI This is required by law ation 901 Federal Avlatio	(49 U.)	S.C. 1421). Failure	Sand AC43. to report can	9-1 (or subseque result in a civil p	nt revision then enalty not to ax	eof) for instru ceed \$ 1000	ictions
	Make GRUMMAN			-	Model AA-1B				
1. Aircraft	Serial No. AA1B-016	4		T	Nationality and Registration Mark N9664L			1,188	
2. Owner		n on registration certificate)	·		Address (As shown on registration certificate) 409 Wyoming Avanue Wilmington,, DE 19809				
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	4. Unit I				it Identification			5. Type	1
Unit		Make		Model		Serial t	No.	Repair	Alteration
AIRFRAME	ME (As described in Item 1 above)								x
POWERPLAN	IT .								
PROPELLER		,		W 111600			The second secon		
APPLIANCE	Type Manufacture	er .		-					
				6. Conformity	-				
A. Agency's Name and Address Shearer Aviation Services, Inc. 398 C South Ramona Road Lebanon, PA 17042 D. I certify that the repair and/or alteration made to the				B. Kind of Agency X U. S. Certified Mechanic Foreign Certified Mechanic Certified Repair Station Manufacturer unit(s) identified in item 4 above and de			C. Certificate	avarea or	
attachments and that the i	nereto have information fu	peen made in accordance in its true	ance w	with the requirement to the best	of my know	43 of the U.S.	Federal Aviati	on Regulat	ions
70(0	4-20	-2007		Signature of Auth	15		,	-	
Pursuant to the	he authority of the Feder	iven persons specifical Aviation Administra	d belo			was inspected		er prescribe	ed by the
Inthon	it. Standards	Manufacturer	x	Inspection Authorizat		Other (Specify		1	
Y	esignee	Repair Station		Person Approved by Canada Airworthines	Transport is Group	1		7	
1 .	al or Rejection	Certificate or Designation No.		Signature of Author	STATE OF THE PERSON NAMED IN COLUMN 2 IS NOT THE PERSON NAMED IN C	ual /			

FAA Form 337 (12-68)

7 '1 0007 '01

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record.

An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed existing original landing light from the cowl. Installed XeVision High Intensity Discharge (HID) light assembly (kit # XV-36-SL) into vacated position. The ballast was mounted on the firewall. The existing switche, wiring and circuit protection were reused. The provided HID wire harness from the ballast to the light was routed directly and secured for as per XeVision installation instructions. Revised aircraft Weight and Balance and Equipment list. The current draw is slightly less than the original landing light so the circuit protection is of proper amp rating.

The following are instructions for continued alrworthiness for this altered airframe:

- 1. Introduction: This installation was accomplished to increase landing light illumination and to increase the service life of the landing light.
- 2. Description: Removed existing landing light and installed XeVision HID light.
- 3. Control Operation: The light is controlled with the existing aircraft landing light switch. There are no special procedures.
- 4. Servicing Information: The components are not field repairable and must be replaced with approved components.
- Maintenance Instructions: This lighting system is to be maintained in accordance with FAR part 43.13. Inspections are to be performed in accordance with FAR part 43.15.
- 6. Troubleshooting Information: If the circut breaker pops, replace the ballast unit. If the light does not Illuminate, remove the lamp or ballast, verify function and replace as required. Bench testing must be done in accordance with the XeVision installation and operation instructions with contain warnings for bench testing.
- 7. Removal and Replacement Instructions: The HID lamp is removed and replaced in the same manner as the original lamp. The ballast is attached to the firewall using standard procedures and hardware. The wire harness between the ballast and the lamp shall be removed in accordace with the XeVision installation and operation and instructions.
- 8. Diagrams: Access is through the removal of the cowl. No diagrams required.
- 9. Special Inspection Requirements: Not applicable.
- 10. Application of Protective Treatments: Not applicable.
- 11. Special Hardware: Not applicable.
- 12. Special Tools: Not applicable,
- 13. Commuter category Aircraft: Not applicable.
- 14. Recommended Overhaul Periods: Not applicable.
- 15. Airworthiness Limitations: There are no additional airworthiness limitations.
- 16. Revisions: To revise these instructions for continued airworthiness, a letter will be submitted to the local Flight Standards District Office with a copy of the revised Form 337 and revised ICA.
- 17. Implementation and Record Keeping: These instructions for continued airworthiness are to be placed in the aircraft permanent records and referred to during aircraft systems inspections and maintenance.

END.